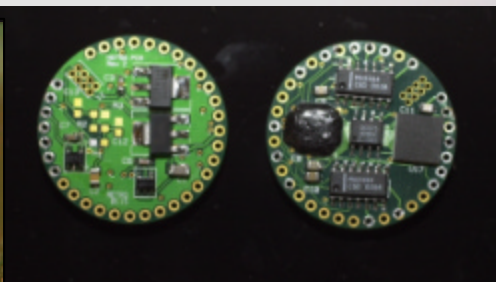
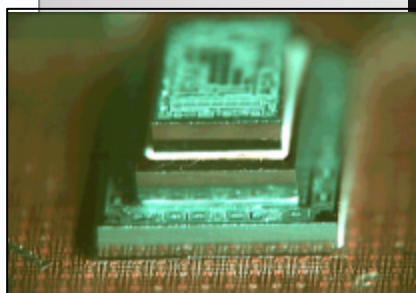


## *Embedded Instrumentation for Smart Munitions*

David H. Lyon

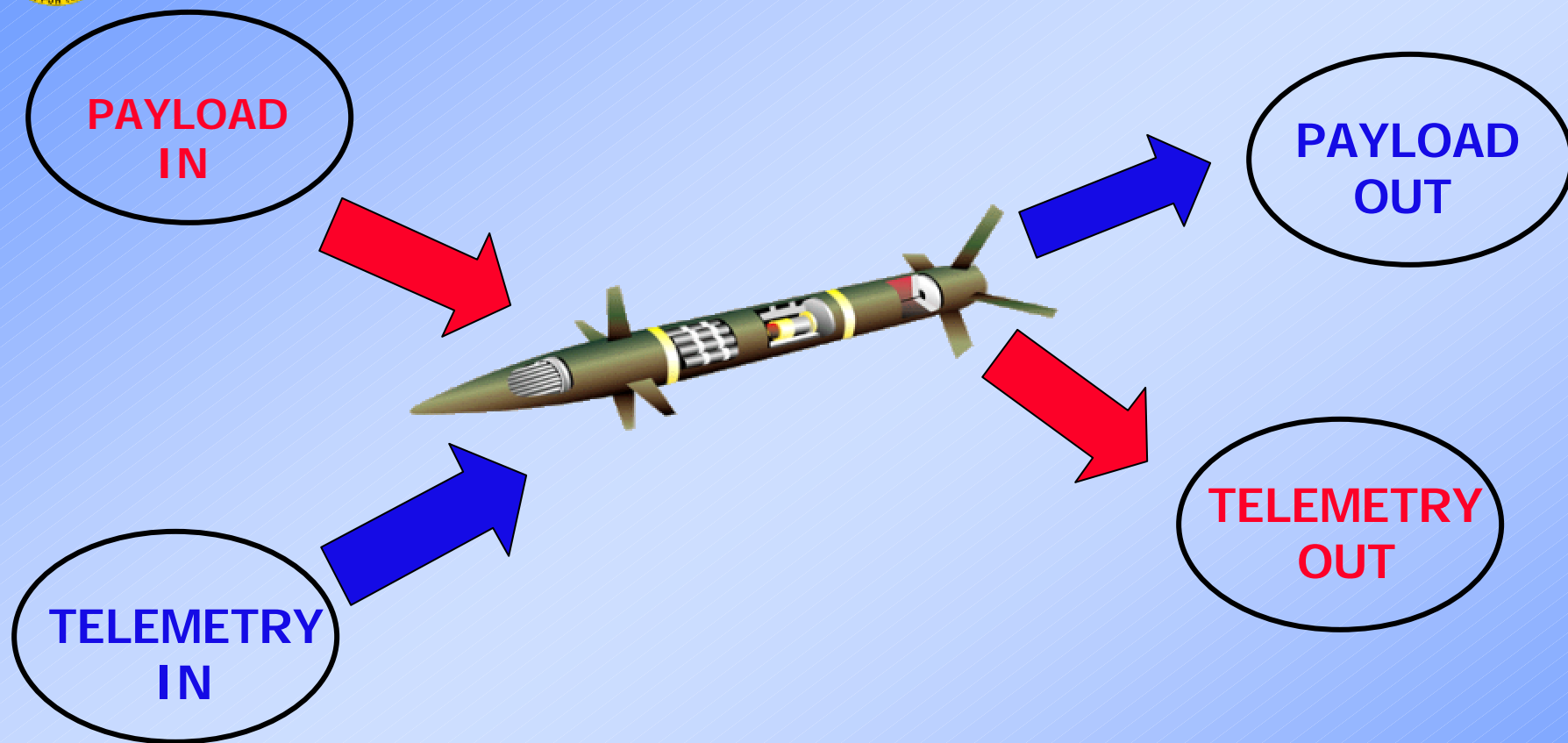
Chief, Advanced Munitions  
Concepts Branch

Weapons and Materials Research  
Directorate  
Army Research Laboratory





# *SOP of the Past*



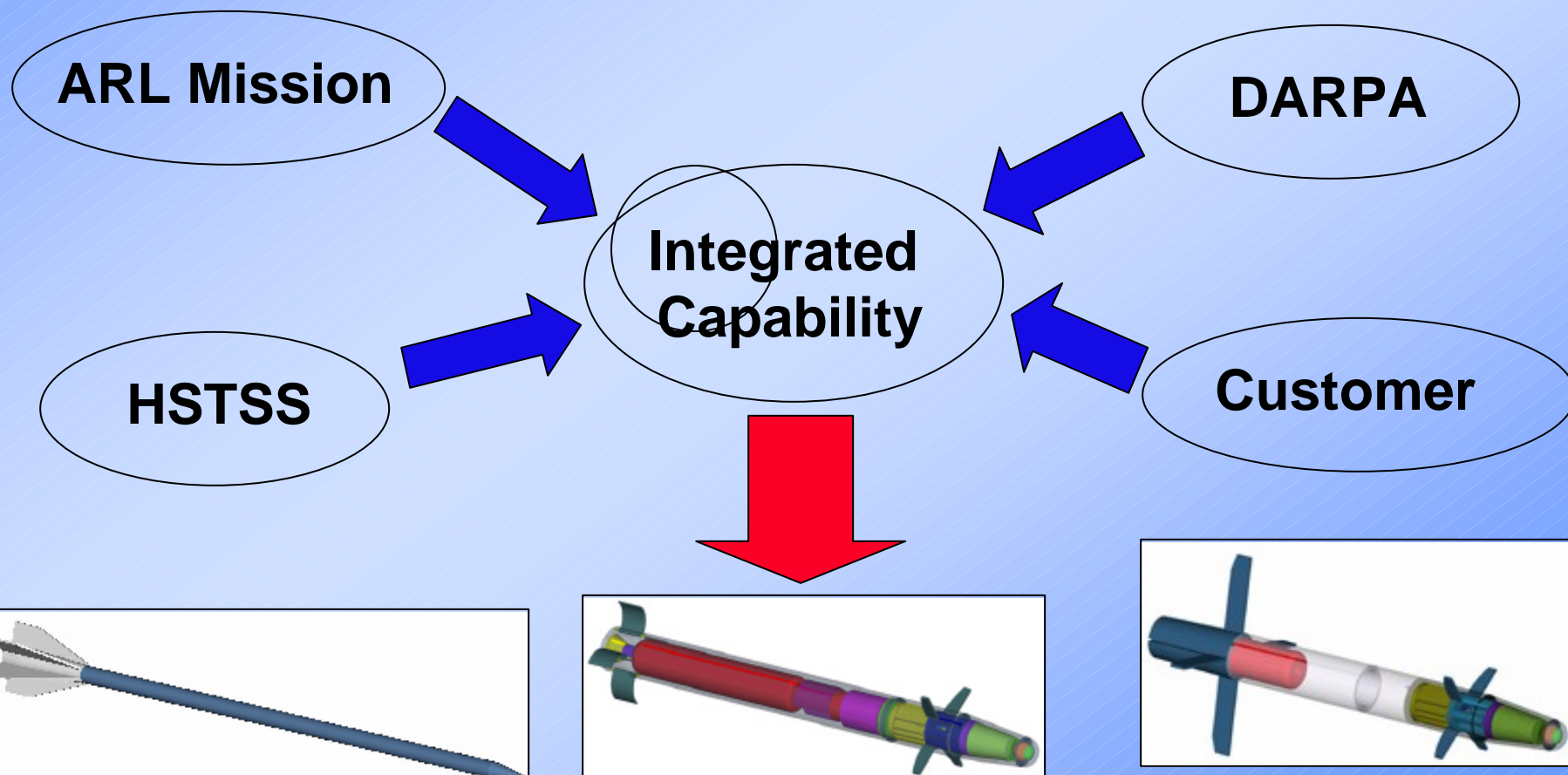
**As we transition from dumb bullets to smart projectiles, the T&E requirements increase tremendously**



# ***New Paradigm***

## ***Embedded Diagnostic Instrumentation***

**Single design, with embedded diagnostics, to cover all environments (cradle-to-grave)**





# ARL Mission

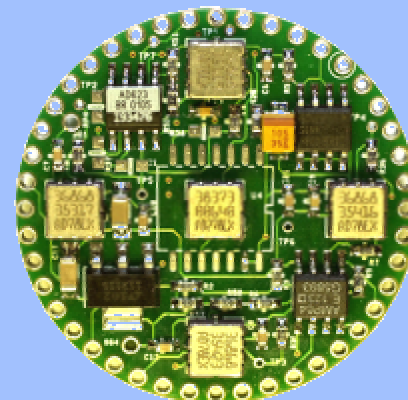
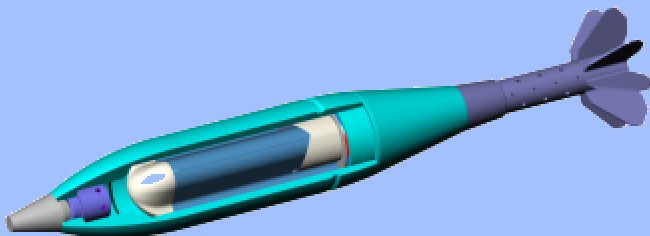
## Advanced G-Hardened G,N&C Suites



### Objectives

*Prototype advanced high-g components for smart munitions:*

- Develop low-cost sensor suites for gun-launch and high spin
- Develop next-generation telemetry components (embedded)
- Demonstrate techniques to measure roll, angular orientation, etc.
- Prototype devices to reduce Target Location Error (TLE)



### Pacing Technologies:

- High-g microelectronics
- Embedded instrumentation
- COTS sensor integration
- Simulation algorithms

### Warfighter Payoffs:

- Increased first round hit
- Slash development time/cost
- Provide ground-truth for T&E
- Validated designs for FCS



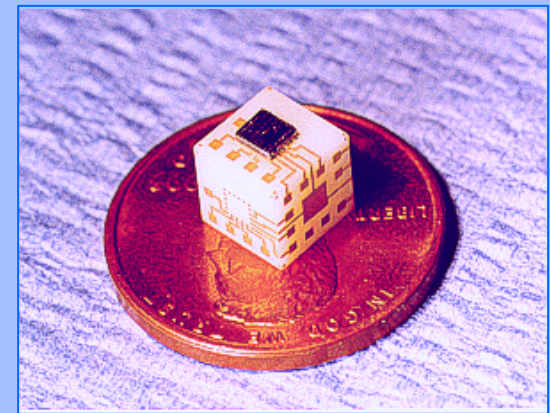
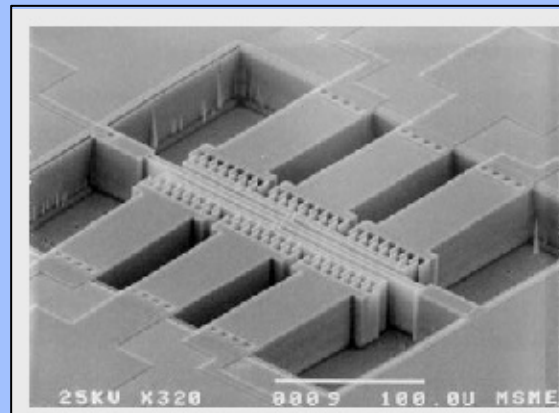
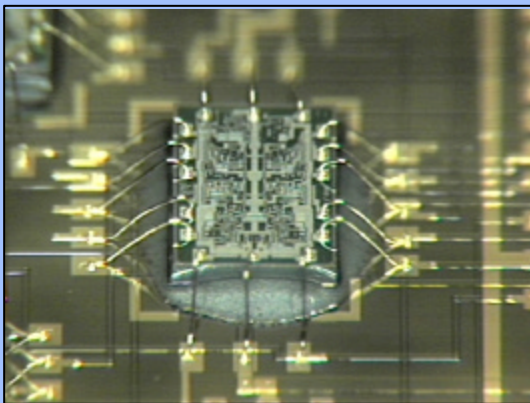


# ***Hardened Subminiature Telemetry and Sensor Systems (HSTSS)***



***Goal:*** Develop advanced instrumentation and packaging for the T&E of high-g, gun-launched smart munitions

***Approach:*** Utilize COTS technology, leverage DARPA investments, use ARL as technical lead and systems integrator



## **Pacing Technologies:**

- Design and procurement of die level comp.
- Advanced packaging technologies - MCM, Chip Stacking, Flip Chip
- MEMS based sensors

## **DoD/Warfighter Payoffs:**

- Lower-cost and lower-risk development cycle for smart munitions
- Embedded diagnostics for seamless transition from R&D, Production, Life Cycle monitoring



# Reference Oscillator



## Statek Crystal Reference Oscillator Requirements

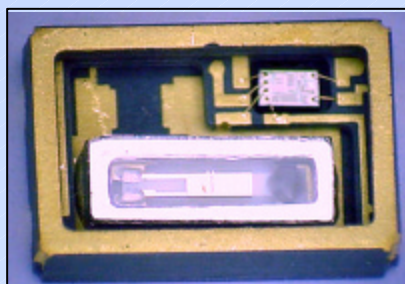
Nominal Output Frequency	20MHz	Acceleration Sensitivity	0.4Hz/G
Frequency Stability	+/- 20ppm	Phase Noise	-140dBc/Hz max at 100kHz
Output	Square Wave	Jitter	250ps max
Supply Voltage	3.0V (+/- 5%)	Rise/Fall Time	8ns max
Max Physical Size	350 x 300 x 150 mils	Duty Cycle	40% - 60%
Operational Temperature Range	-40 to +85degC	Current Draw	5mA max
Shock Ranges	500G, 30kG, 100kG	StartUp Time	10ms max



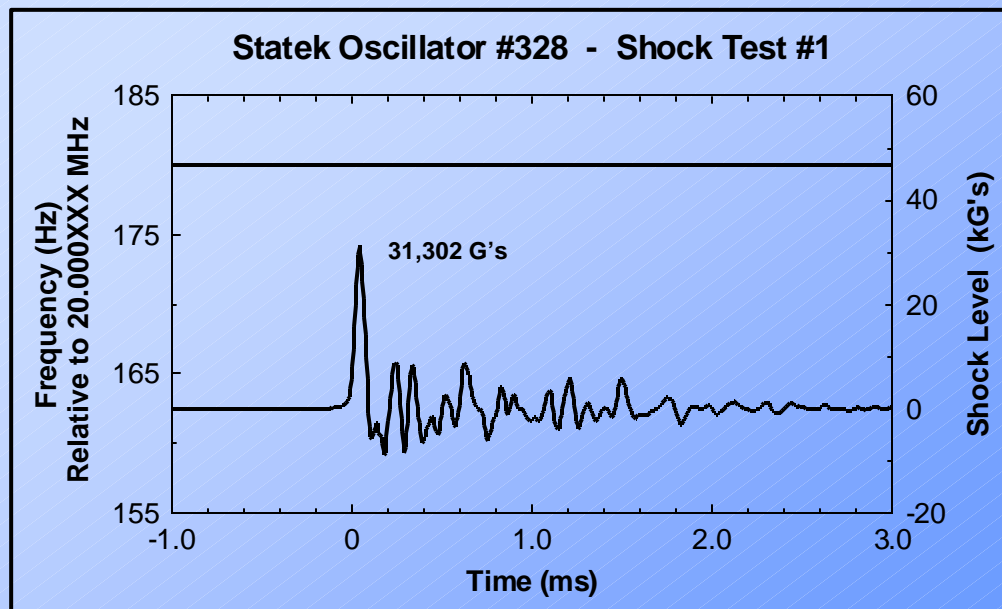
Crystal Resonator



Oscillator

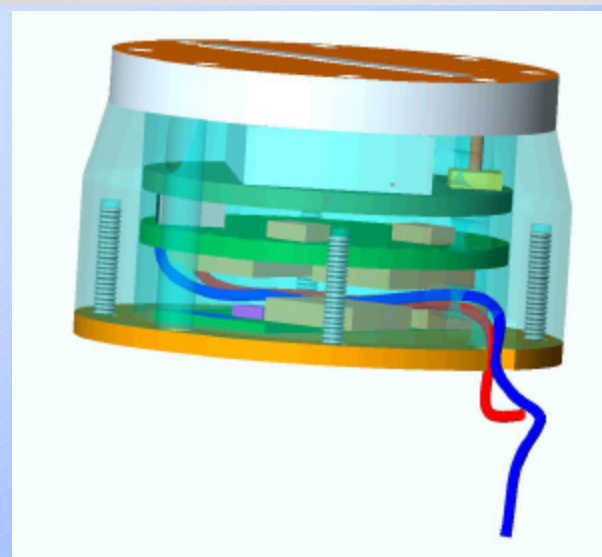
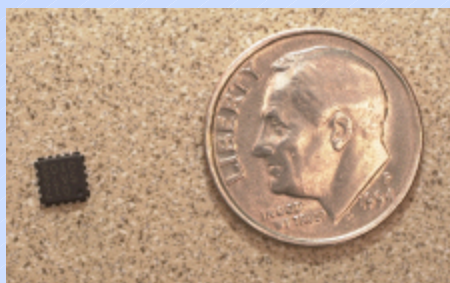
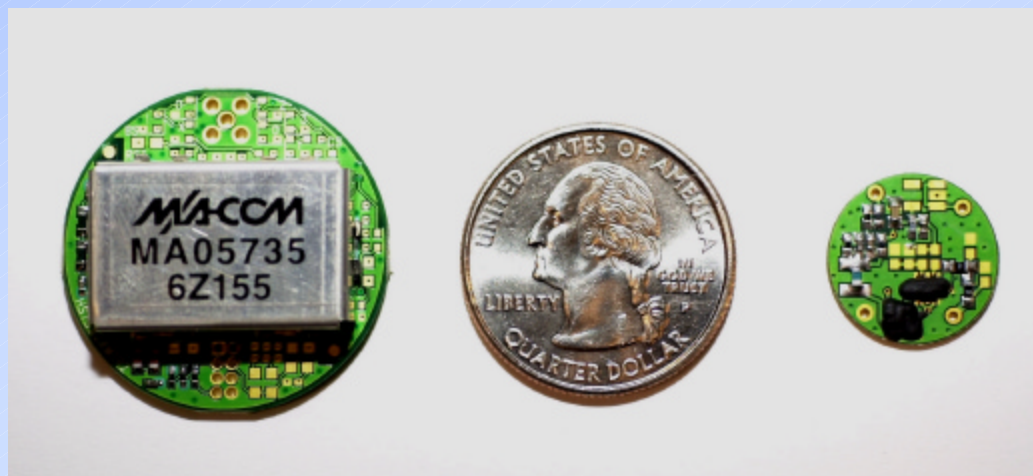
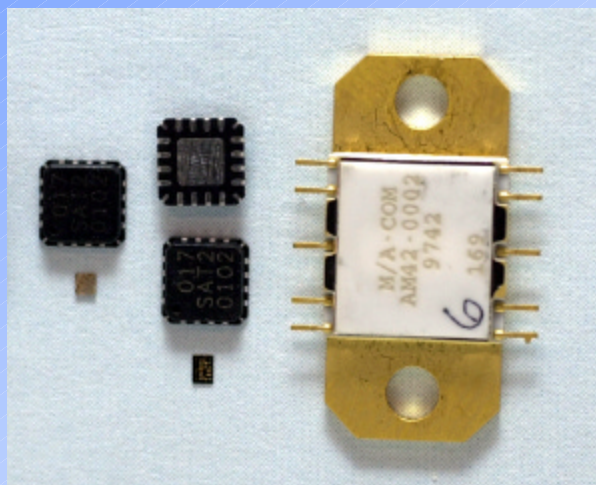


Crystal Resonator Mounted  
Inside Oscillator Package





# Transmitters







# HSTSS Batteries

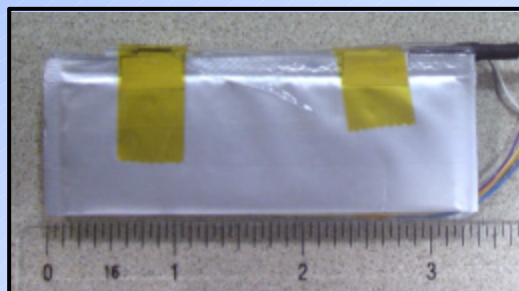
## Ultralife Primary Li/MnO<sub>2</sub> Pouch Cell Batteries

**Artillery Nose Fuze Battery**



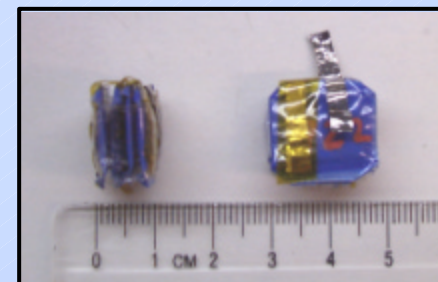
Nominal Voltage: 24V  
 Dimensions: 0.98" x 0.98" x 1.12"  
 Typical Discharge Current: 250mA  
 Minimum Runtime: 30 minutes to 12V

**2.75" Rocket Battery**



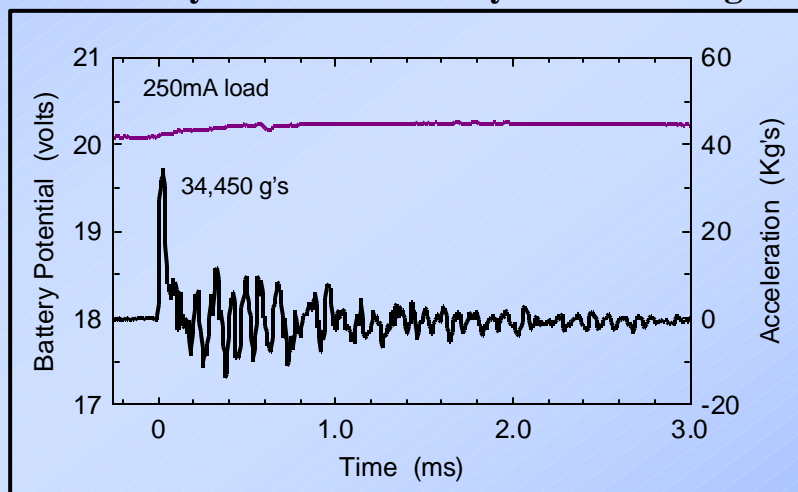
Nominal Voltage: 21V  
 Dimensions: 3.38" x 1.25" x 0.50"  
 Typical Discharge Current: 500mA  
 Minimum Runtime: 2.5 minutes to 12.5V

**KE Projectile Tracer Well Battery**

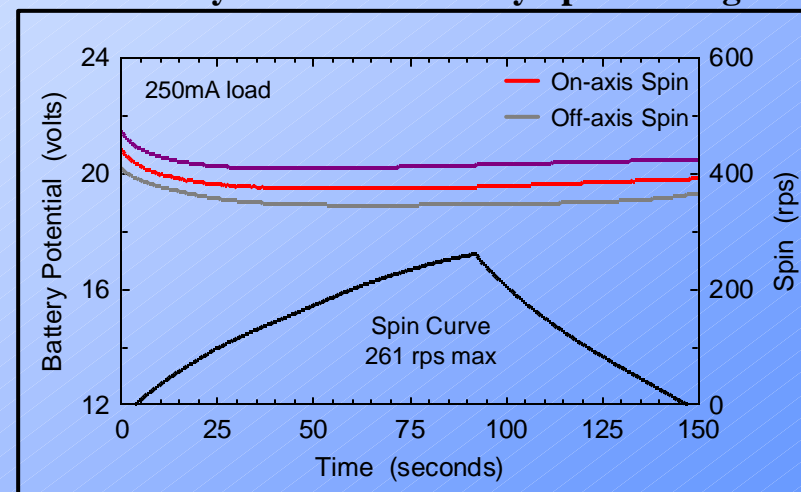


Nominal Voltage: 12V  
 Dimensions: 0.63" x 0.67" x 0.348"  
 Typical Discharge Current: 150mA  
 Minimum Runtime: 6 seconds to 6V

**Artillery Nose Fuze Battery Shock Testing**



**Artillery Nose Fuze Battery Spin Testing**



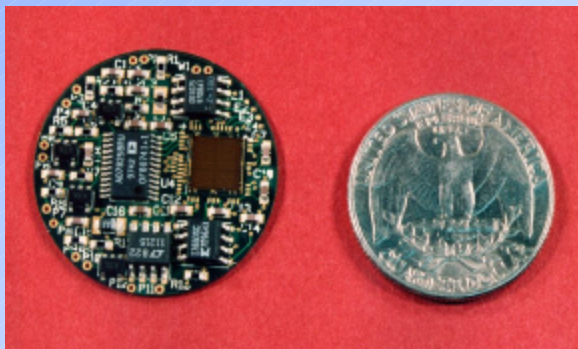




# ***HSTSS Data Acquisition Solutions***

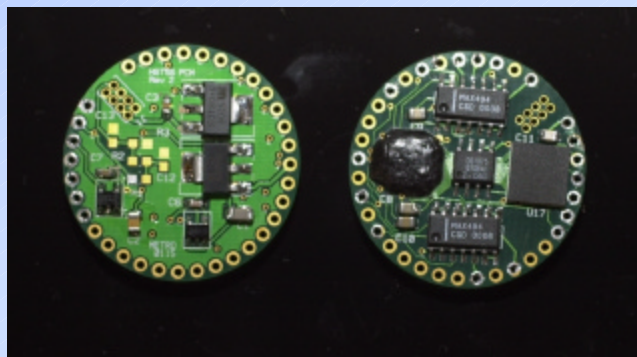


- CPLD based PCM encoders developed at ARL
- FPGA and PIC based PCM encoders developed at NAWC



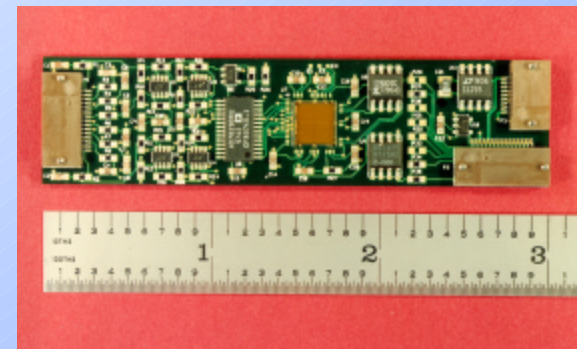
## **4 Channel/ 8 Bit**

- 500 Kbps
- SR = 8.93KHz/ch
- 30 mA @ 5 V



## **8 Channel/ 12 Bit**

- Up to 5 Mbps
- SR = 37.9 KSPS/ch
- 65 mA @ 5 V



## **8 Channel/ 8 Bit**

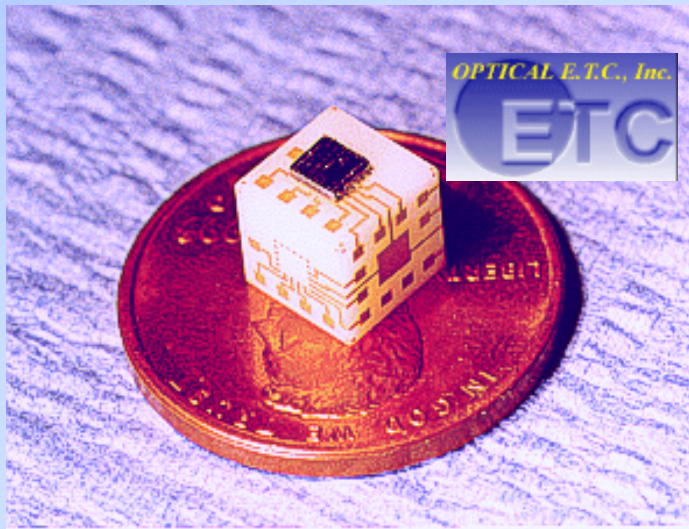
- 240 Kbps
- 1 ch sampled @ 10 KHz
- 7 ch sampled @ 2 KHz



# ***HSTSS Sensor Development***



- Evaluate COTS MEMS technology for ballistic applications
- Develop process for qualifying MEMS devices
- Leverage DARPA & other service investments



## **Optical E.T.C. Inc. Air Force SBIR**

- 3 axis MEMS acceleration package
- 2 accelerometers per chip (dual range)
- Flip Chip design
- Further package development
- Enhance accelerometer performance

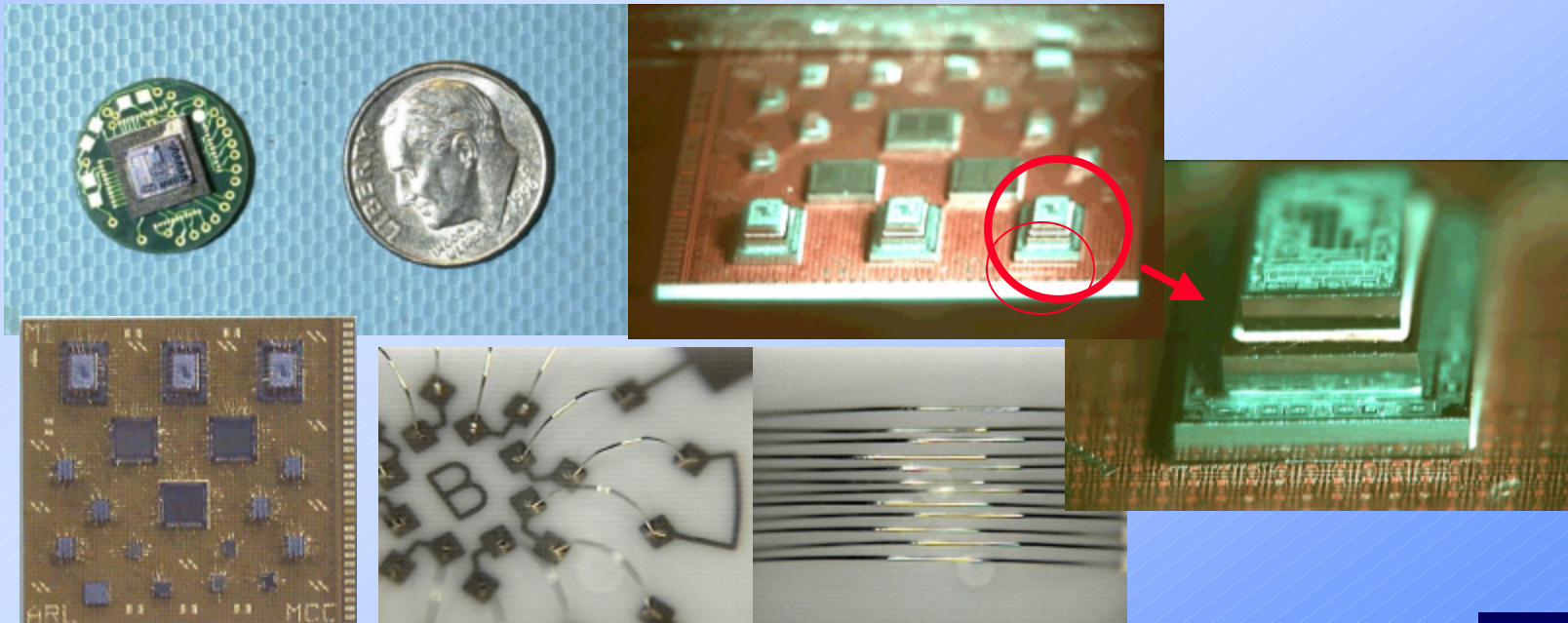


# HSTSS

## *Advanced High-g Packaging*



- Establish & qualify microelectronic assembly techniques for ballistic environments
  - Establish boundaries
  - Adopt & modify commercial techniques
  - Address substrate materials, adhesives, interconnect, etc.



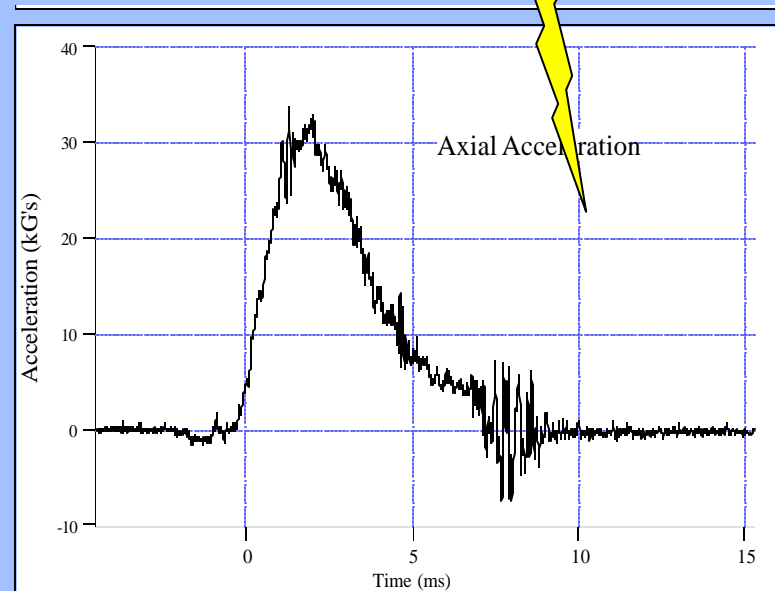
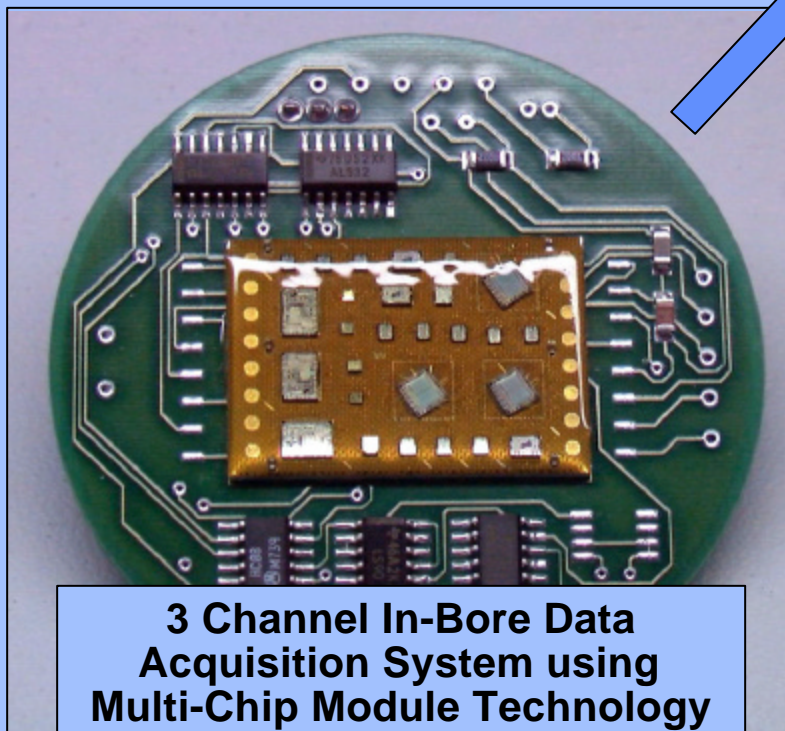
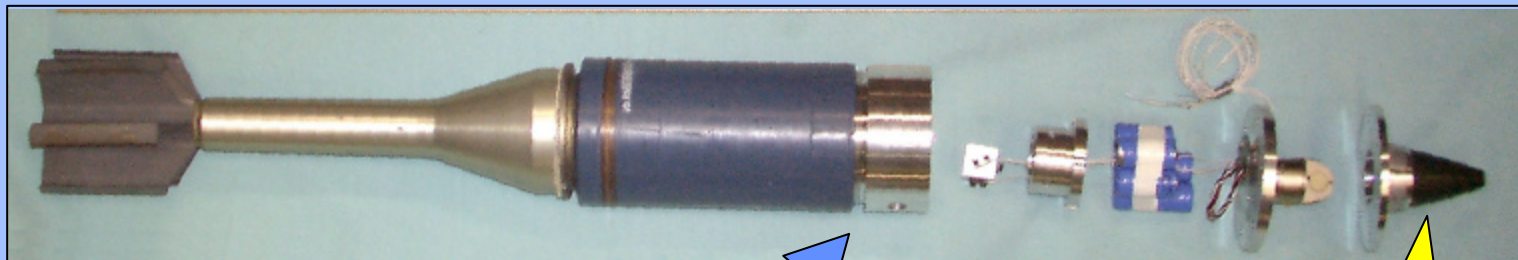




# Very High-G Demonstrations



**Supports FCS and AGS via In-bore and Free Flight Telemetry Capability**



**In-Bore Axial Acceleration**

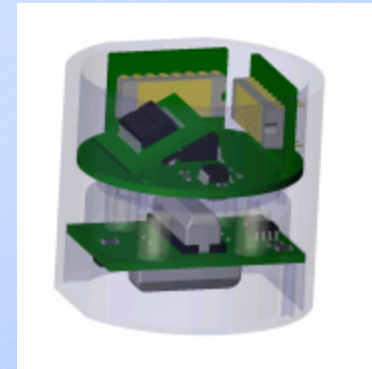
**MCC**







# ***DARPA/ARL Hydra-70 with Impulse Thrusters***



- Integrate MEMS angular rate sensors into IMU to control impulse thrusters
  - TM multi-channel sensor data from angular rate sensors, accels., strain gages, etc.
  - Evaluate devices by comparing similar outputs
  - Simulate trajectory using PRODAS/BOOM



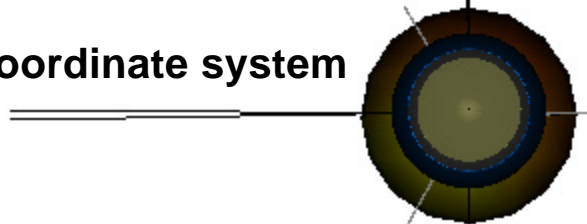
# ***Predicted Hydra-70 Trajectory*** ***Impulse Thrusters***



PRODAS2000 3D - PETRock5\_ver 12-5.pr3

**10 second flight - front view**  
**4 thrusters 0.7, 1.3, 7.0 & 8.0 sec**

**Missile fixed coordinate system**

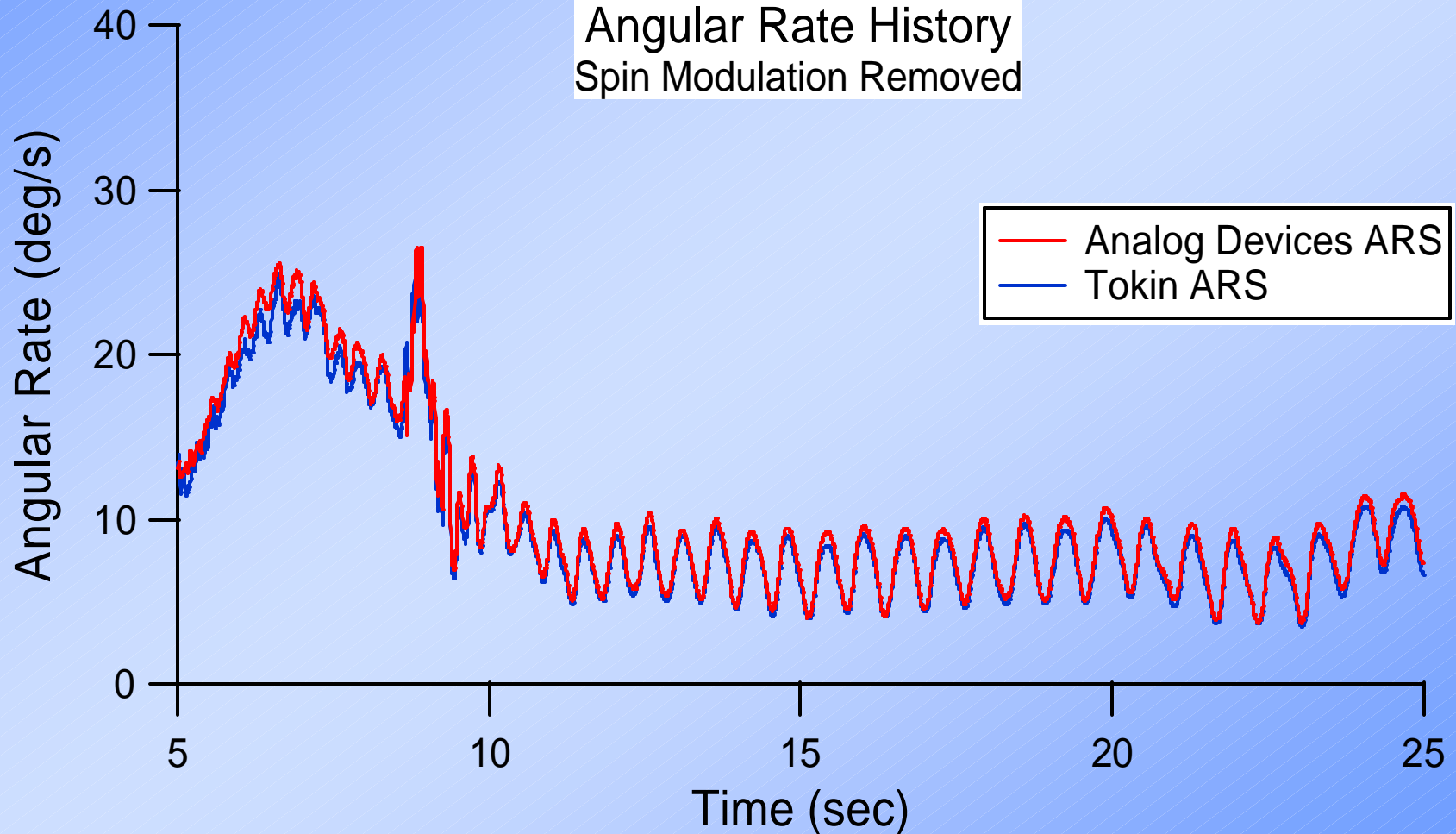


**Earth fixed coordinate system**



## Angular Rate History

### Spin Modulation Removed





# ***DARPA/ARL SCORPION Guided Projectile***



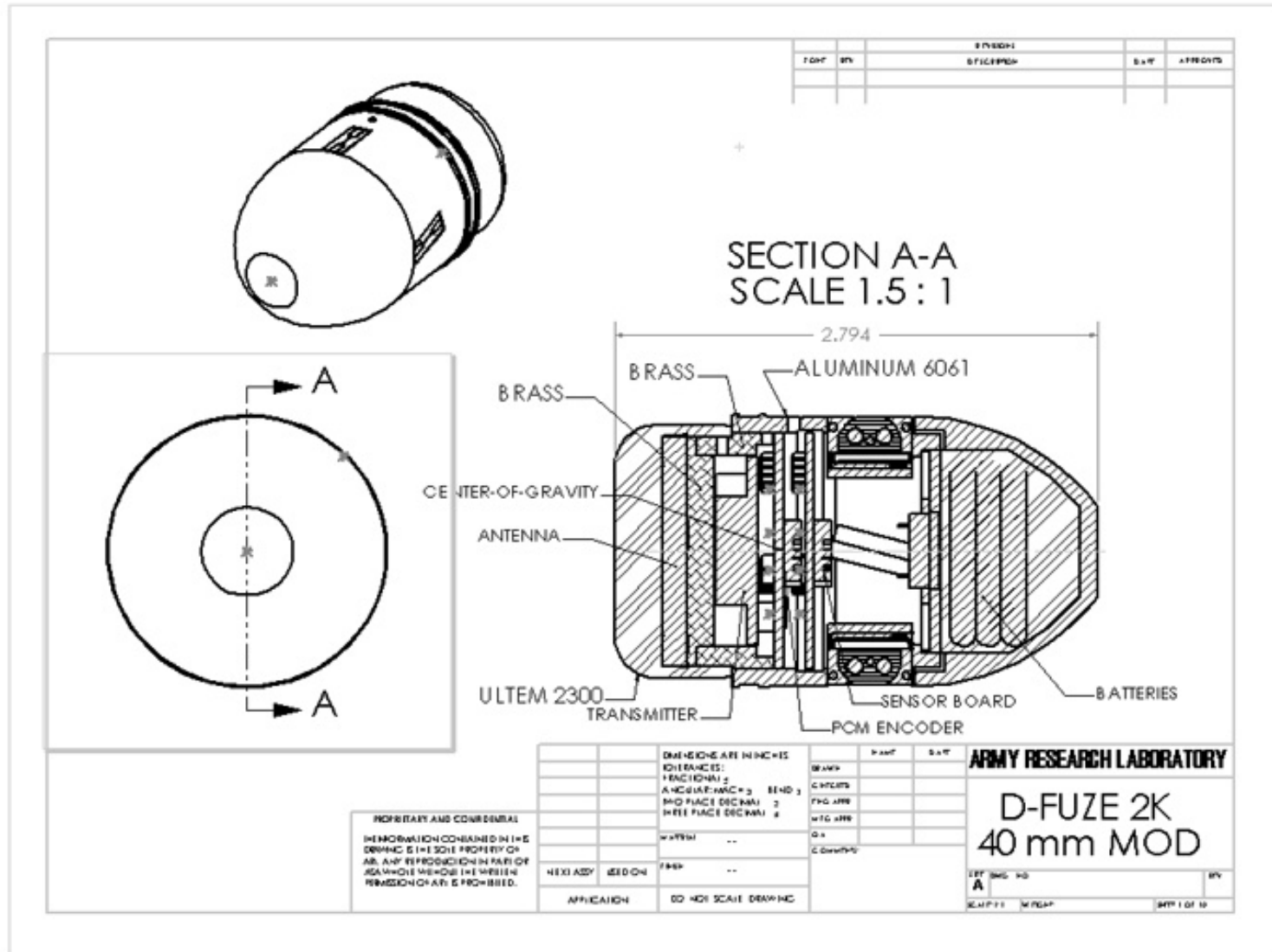
- Integrate G,N&C on medium caliber munition
  - 40-mm grenade utilizing Micro Adaptive Flow Control to provide maneuver authority
  - Provide multi-channel sensor data from surrogate M203 projectile
  - Integrate antenna, transmitter, PCM encoder and battery with sensors
  - Acquire data using ground station





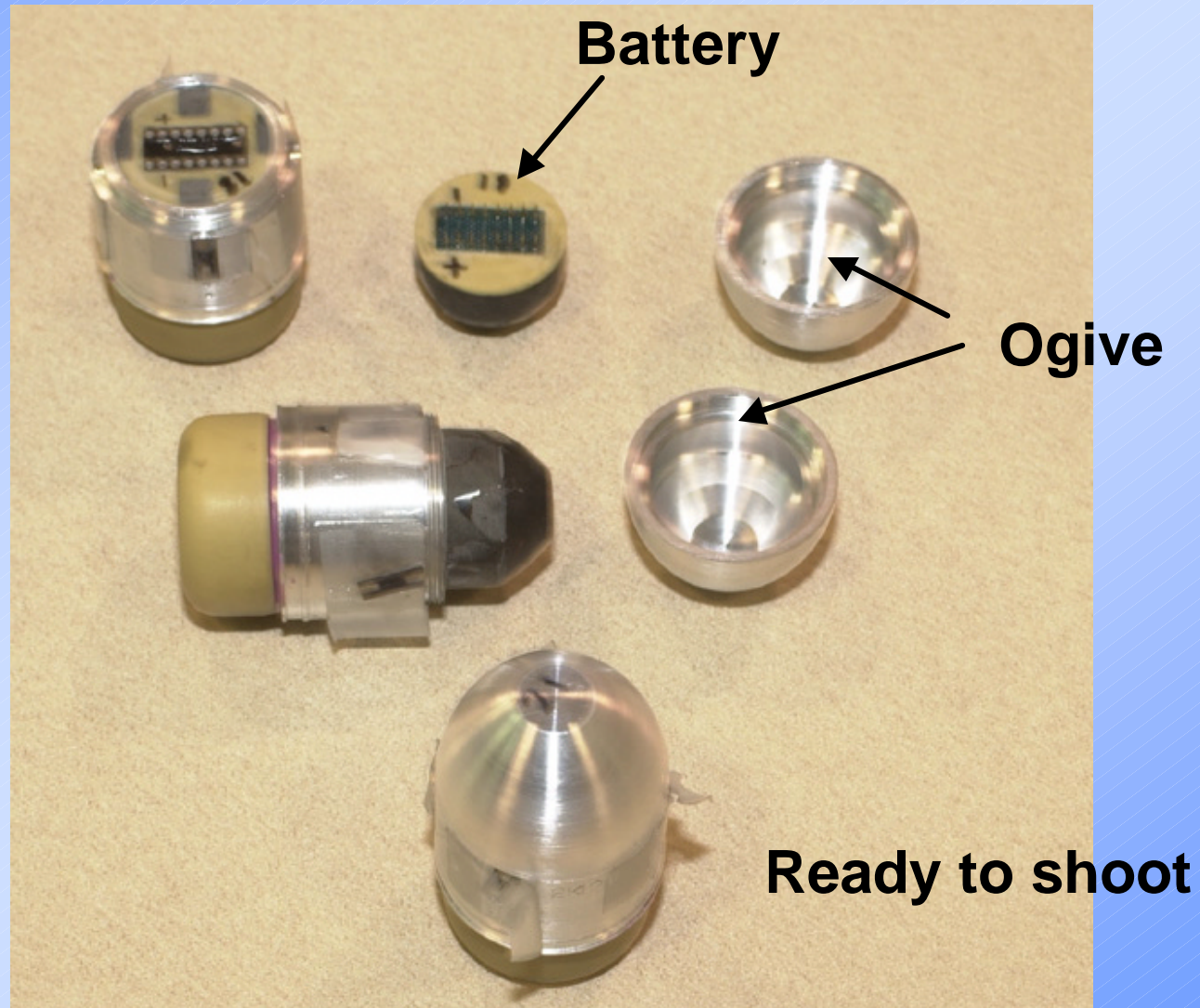


# SCORPION™ PROJECTILE LAUNCH



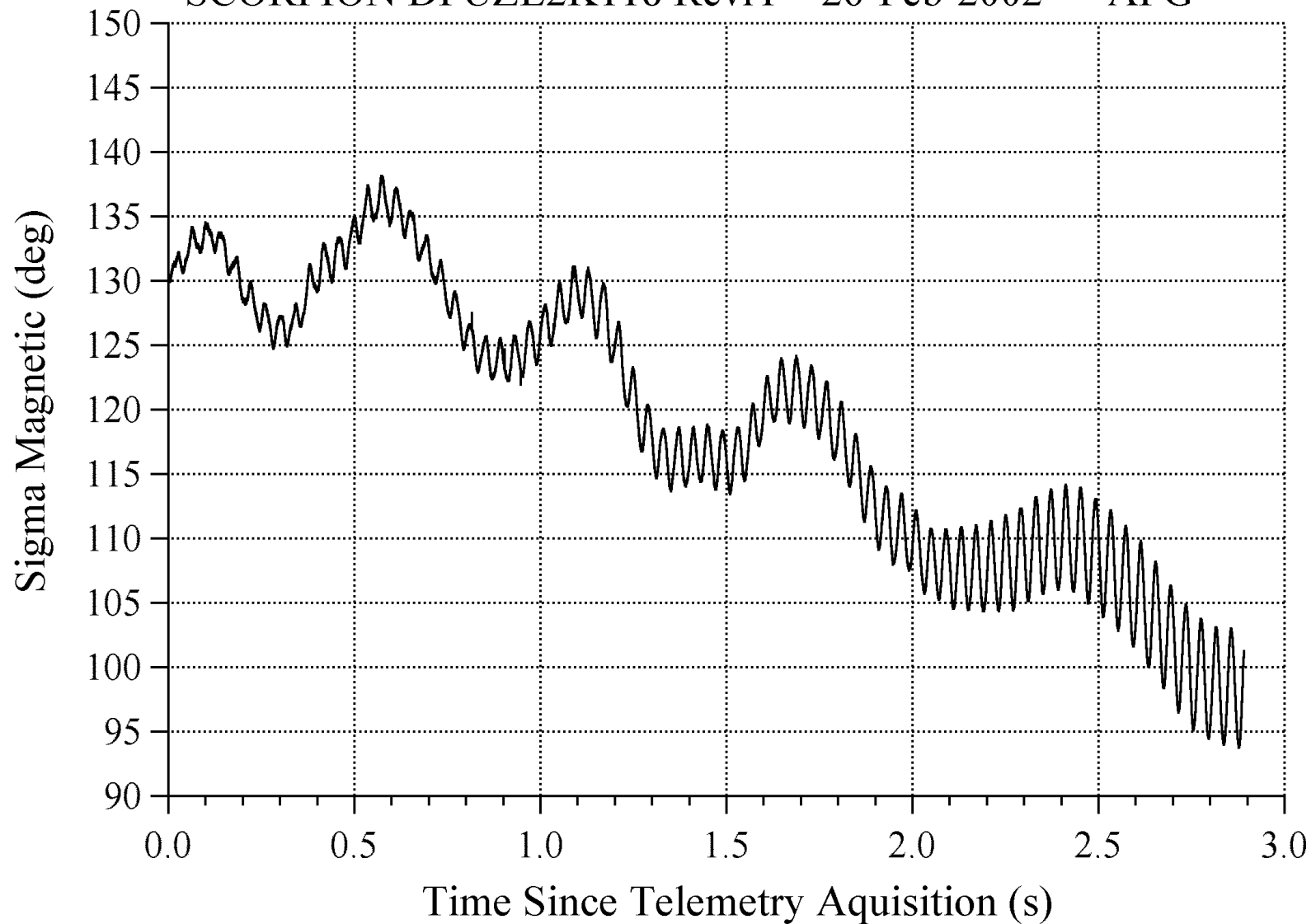


# ***SCORPION TM FLIGHT HARDWARE***



# SCORPION SENSOR SUITE DATA

SCORPION DFUZE2K118 Rcvr1 26-Feb-2002 APG

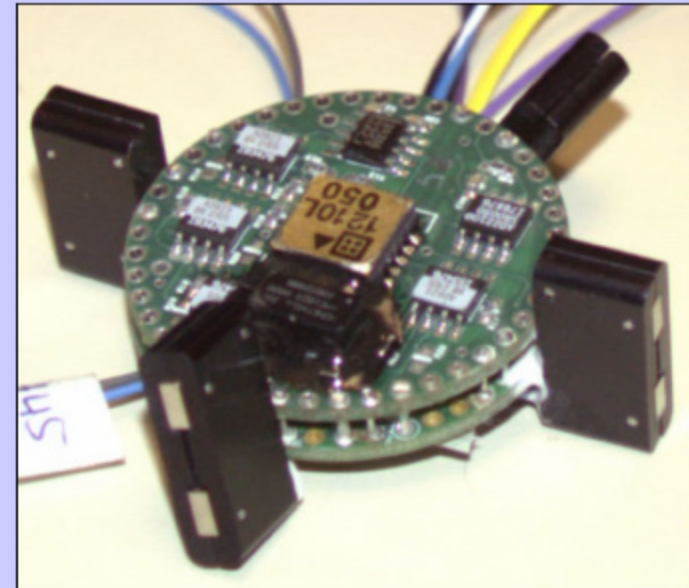
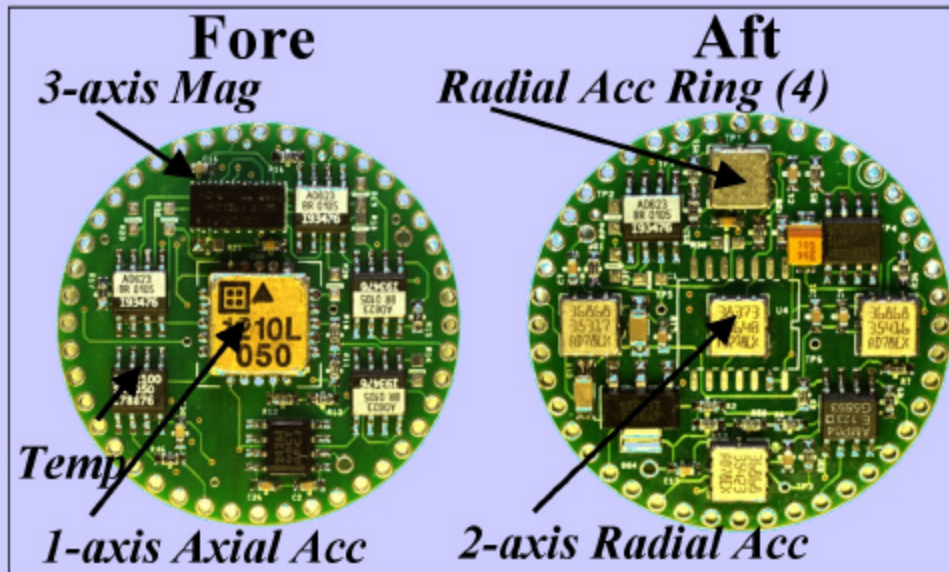




# 40mm Inertial Sensor Suite (ISS)

- 36 mm printed circuit board
- 9 measurement channels

- Uses low-cost parts
- High-g survivable



MEASUREMENT	Signal	PART	MAKER	SELECTABLE RANGES
1-axis Axial Acceleration	Acc I	SD1210	SDI	+/-5, 10, 25, 50, 100, 200, 10k* g's
2-axis Radial Acceleration	Acc J, K	ADXL278	ADI	+/-35, 70 g's
3-axis Magnetic Field	Mag I, J, K	HMC1023	Honeywell	+/-6 Gauss
Accel Ring Spin Rate	Spin	ADXL78 (4 ea.)	ADI	+/-35, 70, 120, 250 g's (0 - 70 Hz)
Solar Field Optical Sensors	Solar	SLIT (4 ea.)	ARL	
Temperature	Temp	AD22100	ADI	-50 to 150 degree C





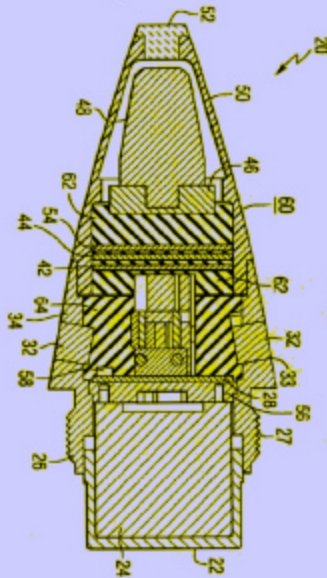
# ARL - Aeroballistic Diagnostic Fuze (DFuze)



- **Problem:** Ground-based instrumentation (i.e. radars, photos, and pressure gages) have limited capabilities.

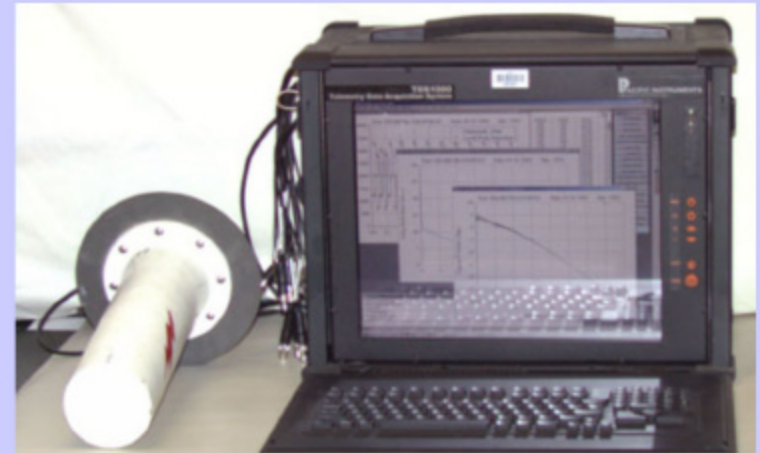
## DFUZE 2001 – High-G Projectile-Borne Instrumentation System

### Artillery Nose Fuse Replacement



(Patent US 6,349,652)

## Portable Data Acquisition System



- Post-Flight processing
  - Quick Look - 6 minutes
  - Final Analysis - 1 month

- **Solution:** DFuze is a projectile-borne, non-intrusive method to verify flight performance, validate aerodynamics and maneuver authority.



# Summary



- Requirements for enhanced capabilities continue to evolve
  - **Embedded diagnostics coupled with telemetry**
- Significant progress through multiple efforts (ARL, HSTSS, DARPA, PMs)
  - **Leveraging approach to maximize return**
  - **Constantly shrinking volume and cost**
- Future efforts center on extending the envelope
  - **Medium and small caliber systems**
  - **EM and ETC launch environments**

